DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION

2H1 Revision 13 BELL 47J 47K 47J-2 47J-2A

February 18, 2005

HELICOPTER SPECIFICATION NO. 2H1

MANUFACTURER Bell Helicopter Textron

Subsidiary of Textron, Inc. Fort Worth, Texas 76101

I - Model 47J, 4 PCLH, Approved August 23, 1956

Engine Lycoming VO-435-A1B, -A1D, or -A1E Fuel 80/87 minimum grade aviation gasoline

Engine limits Maximum r.p.m. 3200 (220 hp.) (See item 602 for 240 hp)
Carburetor & Marvel-Schebler MA4-5, Setting 10-3856 or 10-3856-1 or

carburetor settings Marvel-Schebler MA4-5AA, Setting 10-4025-1

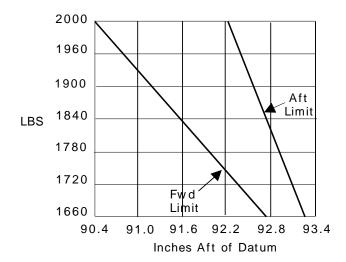
Rotor limits

& operational Power On (Engine Tach.) Power Off (Rotor Tach.)

engine speeds Maximum 3200 Maximum 370 Minimum 3000 Minimum 333

Airspeed limits Never exceed 105 m.p.h. (91.5 knots) True Ind.

C.G. range (+81.4) to (+89.0) Empty weight C.G. range See Figure



Maximum weight 2565 lbs. (See items 6 and 602 for 2800 lbs. and 2850 lbs.)

No. of seats 4. Pilot (+32) to (+35), 3 passengers (+63).

Maximum baggage 250 lbs. (+140)

Fuel capacity

35 gals. (+93) (Usable 34.6 gals.) See NOTE 1 for unusable fuel
Oil capacity

2.9 gals. (+156) (Usable 1.6 gals.) See NOTE 1 for undrainable oil
Rotor blade & control

For rigging information refer to the applicable Bell Maintenance

movements Manual

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Serial Nos. eligible 1420 thru 1443, 1559 thru 1581; 1712 thru 1803

In addition to the pertinent required basic equipment specified in Required equipment

CAR 6, the following items of equipment must be installed: 1, 2, 3, 4, 5, 101, 102(a), 103, 104(a), 105, 201(a), 301(a),

302(a) or (b), 401(a), 402, 601(a), (b) or (c)

II - Model 47K (Navy HTL-7 - See NOTE 4), 2 PCLH, Approved March 30, 1959

Model 47K is similar to the 47J except for cabin, open engine compartment, oil system, lights, and other miscellaneous items in accordance with Bell Report No. 47-947-067. Military Model HTL-7 is similar to Model 47K except for blind flying instruments. See NOTE 4 for modification required for conversion of Navy HTL-7 to Model 47K.

Engine Lycoming VO-435-6A or VO-435-A1D Fuel 80/87 minimum grade aviation gasoline

Engine limits Maximum continuous,

> (Sea level) 24.6 in.Hg., 3200 rpm (220 hp.) (4300 ft.) 23.3 in.Hg., 3200 rpm (220 hp.)

(Straight line manifold pressure variation with altitude) Takeoff (2 minutes) 26.5 in.Hg., 3200 rpm (240 hp.)

Carburetor & Marvel-Schebler MA4-5AA, Setting 10-4025-1

carburetor setting

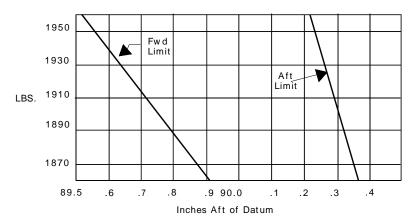
Rotor limits

& operational Power On (Engine Tach.) Power Off (Rotor Tach.) engine limits Maximum 3200 Maximum 370 Minimum 3000 Minimum 333

Never exceed 105 m.p.h. (91.5 knots) True Ind. from S.L. to 7000 ft. Airspeed limits

Above 7000 ft. decrease Vne 4 knots per 1000 ft.

C.G. range (+82.0) to (88.5)Empty weight C.G. range See Figure



Maximum weight 2565 lbs. No. of seats 2(+51.6)

Maximum baggage For baggage loading refer to the applicable Flight Manual Fuel capacity 35 gals. (+93) (Usable 34.6 gals.) See NOTE 1 for unusable fuel Oil capacity 2.7 gals. (+121) (Usable 1.5 gals.) See NOTE 1 for undrainable oil Rotor blade & control For rigging information refer to applicable Bell Service Manual

movements

Serial Nos. eligible 2114 thru 2131

Required equipment In addition to the pertinent required basic equipment specified in

CAR 6, the following items of equipment must be installed:

1, 2, 3, 4, 5, 101, 102(a), 103, 104(a), 105, 201(a), 301(b), 302(a) or (b), 401(b)

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III - Model 47J-2, 4 PCLH, Approved January 14, 1960

Model 47J-2 is similar to the 47J except the basic helicopter incorporates metal main rotor blades, fixed stabilizer, VO-540 engine, blue tinted bubble and windows, and miscellaneous minor improvements. (See Note 7 for conversion of 47J to 47J-2.)

Lycoming VO-540-B1B or -B1B3 (See Item 610 for 260 hp.) Engine

80/87 minimum grade aviation gasoline Fuel

Engine limits Maximum continuous,

> (Sea level) 20.5 in.Hg., 3200 rpm (220 hp.) (9,500 ft.) 19.3 in.Hg., 3200 rpm (220 hp.)

(Straight line manifold pressure variation with altitude) Takeoff (2 minutes) 22.0 in.Hg., 3200 rpm (240 hp.) Marvel-Schebler MA-6AA, Setting No. 10-4218

Carburetor & carburetor settings

Rotor limits

& operational Power On (Engine Tach.) Power Off (Rotor Tach.)

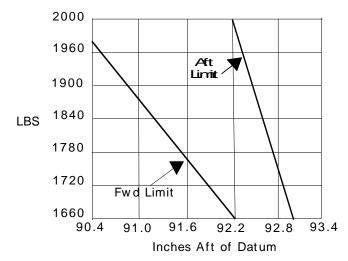
engine limits Maximum 3200 Maximum

> 3000 Minimum Minimum 333

Airspeed limits Never exceed 105 m.p.h. (91.5 knots) True Ind.

C.G. range (+81.4) to (+89.0)

Empty weight C.G. range See Figure



Maximum weight 2850 lbs.

No. of seats 4. Pilot (+32) to (+35), 3 passengers (+63).

Maximum baggage 250 lbs. (+140)

48 gals. (+93) (Usable 47.5 gals.) See NOTE 1 for unusable fuel Fuel capacity Oil capacity 4.3 gals. (+114.5) (Usable 2.5 gals.) See NOTE 1 for undrainable oil Rotor blade & control For rigging information refer to the applicable Bell Maintenance

movements Manual

Serial Nos. eligible 1804 and up and Model 47J converted per Note 7.

Required equipment In addition to the pertinent required basic equipment specified in CAR 6,

the following items of equipment must be installed:

2, 3, 4, 5, 6(a), 101, 102(b), 103, 104(b), 105, 106, 201(a), 301(a), 302(b), 401(c)

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IV - Model 47J-2A, 4 PCLH, Approved March 4, 1964

Model 47J-2A is similar to the 47J-2 except for increased weight, the installation of collective boost system, addition of main rotor tip weights, a change in the C.G. limits, and miscellaneous minor improvements.

Lycoming VO-540-B1B3 Engine

Fuel 80/87 minimum grade aviation gasoline

Engine limits Maximum continuous,

> (Sea level) 20.5 in.Hg., 3200 rpm (220 hp.) (9,500 ft.) 19.3 in.Hg., 3200 rpm (220 hp.)

(Straight line manifold pressure variation with altitude) Takeoff (2 minutes) 23.6 in.Hg., 3200 rpm (260 hp.) Marvel-Schebler MA-6AA, Setting No. 10-4218

carburetor settings

Carburetor & Rotor limits

Power On (Engine Tach.) Power Off (Rotor Tach.) & operational

370 engine limits Maximum 3200 Maximum

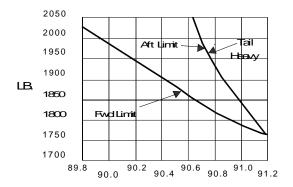
Minimum 3000 Minimum 322

S.L. to 3000 ft. Vne = 105 mph (91.5K)Airspeed limits

Above 3000 ft. decrease Vne 4 mph per 1000 ft.

C.G. range (+81.0) to (+87.7)

Empty weight C.G. range See Figure, or see page 1-30 of Maintenance Manual



Maximum weight 2950 lbs.

No. of seats 4. Pilot (+32) to (+35), 3 passengers (+63).

Maximum baggage 250 lbs. (+140)

48 gals. (+93) (Usable 47.5 gals.) Fuel capacity

See NOTE 1 for unusable fuel

Oil capacity 4.3 gals. (+114.5) (Usable 2.5 gals.)

See NOTE 1 for undrainable oil For rigging information refer to the applicable Bell Service Manual

Rotor blade & control

movements Serial Nos. eligible

3108 and up

In addition to the pertinent required basic equipment specified in Required equipment

> CAR 6, the following items of equipment must be installed: 2(b), 3, 4, 5, 6(c), 101, 102(b), 103, 104(b), 105(b), 106,

201(a) or 210(e), 301(c), 401(d)

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Specifications Pertinent to All Models

Station 0 (87 in. forward of centerline of mast.) Datum Leveling lugs lower left hand longeron aft of mast and adjacent cross tube Leveling means Type Certificate No. 2H1 (CAR effective January 15, 1951, Amendments Certification basis 6-1, 6-2, 6-3, 6-4, 6-5, and 6-6) (Structural loading cond. skid landing gear dated 11-12-74)

Production basis Production Certificate No. 100

A plus (+) or minus (-) sign preceding the weight of an item indicates the net weight change when that item is installed. Equipment:

"- -" indicates "same as preceding model"

" " indicates "does not apply"

"_" indicates "do	es not apply"			
Rotors and Rotor Equipment	47J	47K	47J-2	47J-2A
1. Wood rotor blades, Bell 47-110-401 See NOTE 5.	eligible	eligible		
2. Tail rotor, metal, Bell 47-641-074	eligible			
3. 2 Stabilizer Bar Dampers, Houde Eng.				
Div. A13965	4 lbs. (+87)			
4. Main rotor counterweights, Bell 47-110-239	use actual			
and 61-103-120	wt. (+87)			
5. Main rotor hub, Bell 47-120-184	91 lbs (+87)			
6. (a) Main rotor metal blades, Bell 47-110-250-11 or -21	saa Itam 6(b)		STD	
(b) Main rotor metal blades kit, Bell	see Item 6(b)		SID	
47-706-618, installed in accordance				
with Service Instruction 316SI. Bell				
47-706-618-1	+28 lbs (+83)			
or -5 (includes Item 602). When Item 602	. 20 105 (105)	_	_	_
is installed on helicopter, Bell				
47-706-618-3 or -7 is eligible. When				
this Item (6(b)) is installed, maximum				
weight is 2850 lbs.	+25 lbs (+83)			
Flight Manual Supplement dated June 18,				
1958 required.				
(c) Main rotor metal blades,				
Bell 47-110-250-23				
7. Tail rotor blades 47-641-117 SI428	_		use act	STD
			wt. chg.	wt. chg.
Engine and Engine Aggregation				
Engine and Engine Accessories - Fuel and Oil System				
ruei and On System				
101. Starter, Eclipse - Pioneer 756-22c	19 lbs (+87)			
102. (a) Oil cooler, Bell 47-671-025 (Oil cooler	6 lbs (+80)			
valve included with this cooler)				
(b) Oil cooler, Harrison D-65705			7 lbs (+80)	
103. 2 Fan belts, Bell 47-661-041	1 lb. (+68)			
104. (a) Carburetor Air Filter, Purolator 51845	1 lb. (+100)			_
(b) Carburetor Air Filter, Fram X-4079	2 lbs (+89)		2 lbs (+100)	
105. (a) Hydraulic pump, N.Y. Air Brake 67B025 (b) Hydraulic pump, Weldon 6010	2 108 (+89)			2 lbs (+90)
106. Long range fuel tank, Bell 47-706-613,	+8 lbs (+100)		STD	2 108 (+90)
installed in accordance with Service	±0 105 (±100)		310	
Instruction 324SI. Flight Manual				
Supplement dated May 21, 1959 required.				
107. Breather Heater Kit, Bell 47-706-658				1 lb (+55)
108. Primer Kit, Bell 47-606-669	_	_		5 lbs (+114)
•	_	_		` '

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		<u>47J</u>	47K	47J-2	47J-2A
Landing					
	2 Skids, Bell 47-500-022	50 lbs (+74)			
(b)	Float landing gear (2565 lbs. max.	use act. wt.			
	wt.), Bell 47-706-533-1, -3, -5, & -7,				
	installed in accordance with Service				
	Instruction 241SI. Flight Manual				
	Supplement dated March 14, 1957 required.				
(c)	Float landing gear (2850 lbs. max.	117 lbs (+76)			
(0)	wt.), Bell 47-706-533-9 & 11	117 105 (170)			
	installed in accordance with Service				
	Instruction 248SI. Flight Manual				
	Supplement dated March 14, 1957				
	required.				
(d)	High skid gear (2850 lbs. max. wt.),	+10 lbs. (+79)			
	Bell 47-706-628-1 installed in				
	accordance with Service Instruction				
	340SI (Eligible only when Item 6				
	is installed). Flight Manual				
	Supplement dated May 8, 1959 required.				
(e)	Float landing gear (2950 lbs. max.				117 lbs. (+76)
(0)	wt.), Bell 47-706-533-11 installed				117 100. (170)
	in accordance with Service Instruc-				
	tion 248SI. Flight Manual dated				
	March 4, 1964 required.				
T-1	1 1 1 P 1 P 1 P 1				
	l and Radio Equipment	20.11 (+97)		20.11- (+97)	
	Generator, Leece Neville 2591G-12 Generator, Bendix Model 30E16-11C	20 lbs (+87)	 40 lbs (+87)	20 lbs (+87)	
	Generator, Leece Neville G-001-2591GA	_	40 108 (+67)	_	20 lbs (+87)
	Battery, Yardley 14XHR15V	14 lbs (+76)			20 103 (107)
	Battery, AN3154	28 lbs (+76)	28 lbs (+76)	STD	STD
	lio Installation	,	, ,		
(a)	AN/ARN-41A (ADF)		32 lbs (+69)		
(b)	ARC-12 UHF (Command)		39 lbs (+150)	_	
	lio, Lear LTR-6 & LTRA-6, installed				
	ccordance with Service Instruction				
	SI Bell 47-706-563-1	22 lbs (+90.5)		22 lbs (+90.5)	
	1 47-706-563-3	25 lbs (+83.0)	_	25 lbs (+83.0)	
	lio Dare, Bell 47-706-599, stalled in accordance with Service	27 lbs (+104)		27 lbs (+104)	
	istruction 284SI.	27 108 (+104)		27 103 (+104)	
	lio, King KY-95, Bell 47-706-670				+12.5 lbs (+45)
	lio, Narco Mark XII, Bell 47-706-622				+14.7 lbs (+42)
	Equipment				
401. (a)	"Bell Helicopter Model 47J Appv'd				
(h)	Flight Man." dated Aug. 17, 1956. Bell Helicopter Model 47K Appy'd				
(0)	Flight Man." dated Mar. 26, 1959.				
(c)	"Bell Helicopter Model 47J-2 Appv'd				
	Flight Man." dated Jan. 13, 1960.				
(d)	"Bell Helicopter Model 47J-2A Appv'd				
100 4	Flight Man." dated Mar. 4, 1964.	2.11 (. 1.1)		211 (11)	
402. Ant	i-fogging fan, Bell 47-733-063	3 lbs (+11)	_	3 lbs (+11)	

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	457	4577	457.0	457.04
402 () C 1: 1 (D II 47 706 556	47J	47K	47J-2	47J-2A
403. (a) Cabin heater, Bell 47-706-556, installed in accordance with Service	+22 lbs (+92)		+22 lbs (+92)	
Instruction 298SI. Flight Manual				
Supplement dated March 14, 1957				
required.				
(b) Heater installation, Bell 47-706-632				7 lbs (+70.2)
404. (a) Fire Extinguisher, Bell 47-706-549	6 lbs (+23)		6 lbs (+23)	
installed in accordance with Service				
Instruction 294SI.				
(b) Fire Extinguisher, Bell 47-706-651	—			6 lbs (+23)
405. Safety belt, seat, Bell 47-706-604, installed in accordance with Service	negligible		negligible	
Instruction 305SI.				
Miscellaneous Equipment				
601. Fixed ballast (total weight to be				
determined for each helicopter) (a) Bell 47-260-015	(+270)			
(a) Bell 47-200-013 (b) Bell 47-300-005	(+8.5)		(+8.5)	
(c) Bell 47-260-018	(+304)	_		
(d) Bell 47-314-034		(+8)		
602. 2800 lbs. gross weight and 240 hp. kit,	+3 lbs (+42)	<u> </u>		
Bell 47-706-605, installed in accordance				
with Service Instruction 278SI Flight				
Manual dated June 7, 1957 required.				
Engine limits Maximum continuous,				
(Sea level) 24.6 in.Hg., 3200 rpm (220 hp.)				
(4300 ft.) 23.3 in.Hg., 3200 rpm (220 hp.)				
(Straight line manifold pressure variation				
with altitude)				
Takeoff (2 minutes),				
26.3 in.Hg., 3200 rpm (240 hp.)				
603. Night Flying installation, Bell 47-706-515,				
installed in accordance with Service Instruction 297SI. Flight Manual Supplement dated				
December 28, 1956 required.	10 lbs (+60.5)		10 lbs (+60.5)	
604. (a) Rotor brake installation (Manual	7 lbs (+76)		7 lbs (+76)	
operation), Bell 47-706-538,	, 105 (1,70)		, 105 (1,70)	
installed in accordance with Service				
Instruction 289SI. Flight Manual				
Supplement dated February 1, 1957				
required.	0.11 (.70)		0.11 (.70)	
(b) Rotor brake installation (Power operation), Bell 47-706-587, installed	9 lbs (+78)		9 lbs (+78)	
in accordance with Service Instruction				
255SI. Flight Manual Supplement dated				
July 30, 1958, required.				
605. Litter installation, Bell 47-706-517,				
installation in accordance with Service				
Instruction 239SI. Flight Manual				
Supplement dated February 1, 1957				
required. Including litter	+59 lbs (+58)		+59 lbs (+58)	
Less litter	+39 lbs (+38) +22 lbs (+55)		+22 lbs (+55)	
606. Deleted	. == 135 (155)		. 22 100 (100)	

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47Ј	47K 47J-2	47J-2A
607. Hoist installation (Eligible only when +64 lbs	s (+65) +64 lbs (+65)	
helicopter is certified under CAR 8.		
Hoist must be rendered inoperative when		
helicopter is used under standard certi-		
ficate), Bell 47-706-513, installed in		
accordance with Service Instruction		
292SI. Flight Manual Supplement dated		
March 14, 1957, required.		
608. Cargo deck, Bell 47-706-554, installed 27 lbs	(+56) <u>27 lbs (+56)</u>	
in accordance with Service Instruction		
296SI.		
609. Cabin step, Bell 47-369-029, installed negligible	ble negligible	
in accordance with Service Instruction		
327SI.	15.11 (05)	
610. 260 takeoff horsepower kit. Bell Kit No.	+15 lbs (+87)	
47-1603-1 installed in accordance with		
Service Instruction 386SI. Bell		
Helicopter Model 47J-2 Approved Flight		
Manual dated January 13, 1960 (Item		
401c), revised March 21, 1961 required.		
Engine, Lycoming VO-540-B1B3		
Engine limits		
Maximum continuous,		
(sea level) 20.5 in.Hg., 3200 rpm (220 hp.) (9500 ft.) 19.3 in.Hg., 3200 rpm (220 hp.)		
(Straight line manifold pressure variation		
with altitude)		
Takeoff (2 minutes), 23.6 in.Hg., 3200 rpm		
(260 hp.)		
Transmission Bell, P/N 47-620-600-25, required		
with this rating. (Included in kit)		
611. Door ejection, Bell 47-706-748 installed	+28) 1 lb (+28)	
in accordance with Service Instruction		
317SI.		
	(+43.5) 5 lbs (+43.5)	
instld, in accordance with Serv. Inst. 351SI.	. , _	

NOTE 1. Current weight and balance report, including list of equipment included in certificated weight empty and loading instructions, when necessary, must be in each helicopter at the time of original certification and at all times thereafter. In order to obtain the most consistent weight and balance results, all model helicopters should be weighed on hjackpoints rather than on skids or floats. The applicable Bell Erection and Maintenance Manual contains complete information on this subject.

The certified weight empty and corresponding center of gravity location must include the following:

Model 47J -	Unusable fuel (Included in fuel capacity) Undrainable oil (Included in oil capacity)	2 lbs. (+100) 2 lbs. (+100)
Model 47K -	Unusable fuel (Included in fuel capacity) Undrainable oil (Included in oil capacity)	2 lbs. (+100) 2 lbs. (+100)
Model 47J-2 -	Unusable fuel (Included in fuel capacity) Undrainable oil (Included in oil capacity)	2 lbs. (+100) 2 lbs. (+100)
Model 47J-2A -	Unusable fuel (Included in fuel capacity) Undrainable oil (Included in oil capacity)	2 lbs. (+100) 2 lbs. (+100)

NOTE 2. This helicopter shall be operated in compliance with the operating limitations specified in the FAA Approved Helicopter Flight Manual.

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- NOTE 3. Information essential to the proper maintenance of the helicopter including retirement time of critical parts is contained in the applicable Bell Maintenance Manual provided with each helicopter. These values of retirement or service life cannot be increased without FAA Engineering Approval. See NOTE 6.
- NOTE 4. Prior to civil Certification, Navy Model HTL-7 Helicopter must be modified in accordance with Bell Report No. 47-099-013.
- NOTE 5. Wooden main rotor blades 47-110-401-13 are not eligible for use. See Bell Service Bulletin No. 47-04-1-74-1.

NOTE 6. Section 1 of the pertinent Model Maintenance and Overhaul Information Manual includes the following information. These replacement or service live may be superseded by AD's. See equipment listing for eligibility of certain items on the models.

ITEM		PART NUMBER	REPLACE AT Hours
MAIN ROTOR SYSTEM			
Main Rotor blade (Metal)		47-110-250-11 or -21	5000
Counterweight Bracket		47-110-287-9	200
Drag Brace		47-110-301-1	2500
Drag Brace		47-110-372-1	2500
Pitch Change Horn		47-120-126-5	5000
Main Rotor Yoke		47-120-177-1	2500
(If used at any time with	47J & K wood blade:	s)	
Main Rotor Yoke		47-120-177-1	5000
(With metal blades only)			
Main Rotor Grips		47-120-252-5	300
Main Rotor Grips		47-120-252-1, -7, -11, 113, -115	1200
Gimbal Ring		47-120-014-1 through -21	1200
Gimbal Ring		47-120-014-23	4800
Cimour Iting		17 120 011 23	1000
CONTROL SYSTEM			
Collective Sleeve Assembly		47-150-117-5, -13, or -21	5000
Scissor Lever		47-150-249-1 and -5	5000
Bellcrank		47-724-097-7, or -9	2500
(If used at any time with	wood blades)	17 721 057 7, 01 5	2300
Bellcrank (With metal blades o		47-724-097-7, or -9	5000
Lateral Torque Tube	my)	47-725-852-1	1200
Laterar Torque Tube		47-725-052-1	1200
TAIL ROTOR SYSTEM			
Tail Rotor Extension Tube	47-640-055-3	600	
Tail Rotor Pitch Change	17 010 055 5	47-640-069-1 and -3	600
Shaft Bearings		47 040 00) 1 und 3	000
Tail Rotor Blades		47-642-102	AD 80-10-4
Tail Rotor Blades		47-642-117-1	2500
Tail Rotor Yoke		47-641-126-5	2500
Tall Rotor Toke		47-041-120-3	2300
MISCELLANEOUS			
Fan Belts (Matched Sets)		47-661-041 (All dash numbers)	600
Transmission Shear Screw	47-620-485-1	600	000
Transmission Shear Serew	47-020-405-1	47-620-485-9	1200
Engine Mount		47-620-483-9 47-612-171 (All dash numbers)	2500
Engine Mount		47-612-171 (All dash numbers) 47-612-135-147	2300
Engine Mount			2500
		At 2565 pounds maximum gross weight	2500
		At 2800 to 2850 pounds maximum	600
		gross weight	

NOTE

A 47-612-135-147 engine mount which has been operated at 2565 pounds maximum gross weight for less than 900 hours can continue in service, after installation of 2850 pounds gross weight kit, for not more than 600 additional hours, provided total service time does not exceed 900 hours.

NOTE 7. Bell Model 47J helicopters may be converted to Model 47J-2 in accordance with Service Instruction No. 350 S.I.

-END-